National Aeronautics and Space Administration

#### **Goddard Space Flight Center**

Greenbelt, MD 20771



April 25, 2003

Reply to Attn of:

410

#### MEMORANDUM FOR THE RECORD

# National Environmental Policy Act (NEPA) Compliance for Galaxy Evolution Explorer (GALEX)

#### 1.0 Introduction

The NEPA of 1969, as amended (42 U.S.C. 4321, et seq.), requires Federal agencies to consider the environmental impacts of a project in their decision making process. To comply with NEPA and associated regulations (the Council on Environmental Quality (CEQ) Regulations for Implementing the Procedural Provisions of NEPA [40 CFR Parts 1500-1508] and NASA policy and procedures [14 CFR Part 1216 Subpart 1216.3]), NASA has prepared an Environmental Assessment (EA) for routine payloads launched on Expendable Launch Vehicles (ELV's) from Cape Canaveral Air Force Station (CCAFS) and Vandenberg Air Force Base (VAFB) (Ref: Final Environmental Assessment for Launch of NASA Routine Payloads on Expendable Launch Vehicles from Cape Canaveral Air Force Station, Florida, and Vandenberg Air Force Base, California, June 2002). The EA assesses the environmental impacts of missions launched from CCAFS and VAFB with spacecraft that are considered routine payloads.

Spacecraft defined as routine payloads would utilize materials, quantities of materials, launch vehicles and operational characteristics that are consistent with normal and routine spacecraft preparation and flight activities at VAFB, CCAFS, and the Kennedy Space Center. The environmental impacts of launching routine payloads from VAFB and CCAFS fall within the range of routine, ongoing and previously documented impacts that have been determined not to be significant. Spacecraft covered by this EA meet specific criteria ensuring that the spacecraft, its operation and decommissioning, do not present any new or substantial environmental or safety concerns.

To determine the applicability of a routine payload classification for a mission launched from VAFB and CCAFS and coverage under the NASA routine payload EA, the mission is evaluated against the criteria defined in the EA using the Routine Payload Checklist (RPC).

#### 2.0 Mission Description

GALEX is a small explorer class mission that is part of NASA's Structure and Evolution of the Universe theme. The objective of the mission is to map the history and evolution of the universe. GALEX is an orbiting space telescope that will observe hundreds of thousands of nearby and distant galaxies in ultraviolet (UV) light. Additionally, GALEX will view the stars in our own galaxy, the Milky Way. The observations will help scientists understand how galaxies evolve and change, and the history of star formation. GALEX will perform both imaging and spectroscopy, conducting several types of surveys, some of which will be science "firsts." Data collected by GALEX will populate a large unparalleled archive that will fill in missing pieces of the current cosmological puzzle. The archive will be available to both the science community and the general public. GALEX will complement the capabilities of current observatories, such as HST and FUSE and future ones planned for later this decade.

GALEX will be launched on an air launched Pegasus XL launch vehicle over the Atlantic Ocean in the 1<sup>st</sup> Quarter 2003. The Pegasus will be carried by an L-1011 Stargazer from CCAFS. GALEX will orbit 690 kilometers above the Earth at an inclination of 29 degrees to the equator.

GALEX is a 3-axis stabilized spacecraft. The spacecraft weighs 277.2 kilograms, and measures 1.97 meters in height, 1.12 meters in diameter with solar arrays stowed, and 2.79 meters wide with solar arrays deployed. Electrical power to the spacecraft will be provided by gallium arsenide solar arrays and nickel-hydrogen batteries. GALEX uses a single instrument with state-of-the-art UV detectors.

The components utilized in the GALEX spacecraft and instruments are made of materials normally encountered in the space industry. GALEX does not use any radioactive materials. GALEX does utilize lasers in the Ring Laser Gyros (RLG's) used to assist in Attitude Sensing for the spacecraft. The RLG's are contained within an aluminum box and no laser radiation is emitted outside the unit. GALEX does not carry any pathogenic organisms, nor will GALEX return samples to Earth. Materials/hazards associated with GALEX do not raise any environmental or health (safety) concerns.

The results of the GALEX Orbital Debris Assessment indicate that the mission meets the less than 8 m² casualty area requirement, but exceeds the less than 25-year orbit life requirement. Based on Code Q's recommendation, Code S has accepted the Orbital Debris Assessment. (Note: This is an on-orbit issue and not a NEPA issue and does not affect the mission qualifying as a routine payload.)

#### 3.0 NASA Routine Payload Determination

The GALEX mission has been evaluated against the NASA routine payload EA for launches from CCAFS and VAFB, using the RPC (see enclosed Evaluation Recommendation Package). The evaluation indicates that the mission meets the criteria for a routine payload. The mission does not present any unique or unusual circumstances

that could result in new or substantial environmental impacts. Based on this review, it is determined that the GALEX mission qualifies as a routine payload and falls within the scope of the reference routine payload EA. No additional NEPA action or documentation is required.

W. F. Joursend for A. V. Diaz

Enclosure

### **EVALUATION RECOMMENDATION PACKAGE**

Record of Environmental Consideration Routine Payload Checklist NEPA Environmental Checklist

## RECORD OF ENVIRONMENTAL CONSIDERATION

1.	Project Name: Galaxy Evolution Explorer (GALEX)
2.	Description/location of proposed action:  GALEX is an orbiting space telescope that will observe galaxies in ultraviolet light across 10 billion years of cosmic history. GALEX will be launched on an air launched Pegasus XL. The Pegasus will be carried from CCAFS by an L-1011 Stargazer to the launch site over the Atlantic Ocean.
	Date and/or Duration of project: Launch 1 <sup>st</sup> Q 03
3.	It has been determined that the above action:
X	a. Is adequately covered in an existing EA or EIS.  Title: Final Environmental Assessment for Launch of NASA Routine Payloads on ELVs from CCAFS, Florida and VAFB, California  Date: June 2002
**************************************	<ul> <li>b. Qualifies for Categorical Exclusion and has no special circumstances which would suggest a need for and Environmental Assessment.</li> <li>Categorical Exclusion:</li> </ul>
	c. Is exempt from NEPA requirements under the provisions of:
	d. Is covered under EO 12114, not NEPA.
····	e. Has no significant environmental impacts as indicated by the results of an environmental checklist and/or detailed environmental analysis.  (Attach checklist or analysis as applicable)
	f. Will require the preparation of an Environmental Assessment.
<del></del>	g. Will require the preparation of an Environmental Impact Statement.
**************************************	h. Is not federalized sufficiently to qualify as a major federal action.    All   Montant   3/5/03     NEPA Coordinator, Code 205.2   Date     Project Manager, Code 4/10   Date     Date   Date   Date     Date   Date   Date   Date     Date   Date   Date   Date     Date   Date   Date   Date     Date   Date   Date   Date   Date     Date   Date   Date   Date   Date     Date   Date   Date   Date   Date   Date     Date   Date   Date   Date   Date   Date   Date     Date   D

NASA Routine Payload Checklist (1 of 2)

PRO	DJECT NAME:	Gale	X	ayloud Onecklist ( )	,			
	DJECT CONTACT		r. Frank Snow		OF LAUNCH:	3/25/03		
			V. I TORRIC ORIOW	PHONE NUMBER:			410	·
PRC	JECT START DA	NTE: 1	12/5/97 Phase A	PROJECT LOCATION:	<ul> <li>Launch Site</li> </ul>	Explorer's O		
			Perform 1 <sup>st</sup> Ultravio	let All-Sky Imaging Survey	and Map Histo	rv of Star For	nation	. Ovo
PRO	JECT DESCRIP	TION:	Redshift Range 0<2	7<2		· , o o o o o o o o o o o o o o o o o o	Hation	OVE
A.	SAMPLE RET	URN:					T	
	1. Would	the ca	andidate mission retu	ırn a sample from an extrat	errestrial hady	<u> </u>	YES	
B.	RADIOACTIVE	SOUR	CES:		corestrar body		ļ	X
	1. Would	the ca	andidate spacecraft o	carry radioactive materials?			YES	
	2. If Yes,	would	the amount of radio	active sources require laun	ch approval at	the NIACA	-	X
	Manua	1)?	ministrator level or n	ilgner according to NPG 87	15.3 (NASA Sa	afety		
Р	rovide a copy	of the	Radioactive Materia	als Report as per NPG 8715	5.3 Section 5.5	.2.	<u> </u>	
C.								
<u>U.</u>	LAUNCH AND		TOTO A SAN AND A				YES	NO
	Would combin	ine cai ation d	ndidate spacecraft b	e launched using a launch cated in Table 1 below?	vehicle/launch	complex		X
	2. Would	the pro	nosed mission caus	cated in Table 1 below?				
	to exce	ed the	· launch rate approve	e the annual launch rate fo ed or permitted for the affec	r a particular la	aunch vehicle		X
Com	ments:			parameter in the difference	ica laarion sile	· ·		J
D.	FACILITIES:						YES	NO
	<ol> <li>Would to modified</li> </ol>	he car ation o	ndidate mission requ of existing facilities?	ire the construction of any i	new facilities o	r substantial	10	X
				ed been listed as eligible or	listed as histor	-i II		
	Significa	uu.			iisteu as nistor	ically		
Provi	de a brief des	criptio	n of the construction	or modification required:				L
E.	HEALTH AND S	A ####						
1.							YES	NO
١.	radio freque	andiga nev fra	ate spacecraft utilize	any hazardous propellants	, batteries, ord	nance,		
**************************************	exceeding [		velope rayload chan	other subsystem component acteristics (EPCs) in Table	2 halow2	i	***************************************	Х
2.	VVould the c	andida	ate spacecraft utilize	any potentially hazardous	matarial as	t of a flight		
	Cyclein will	SO LYP	e or amount brecind	es acquisition of the neces	ean/ narmite nr	ior to its use		х
3.	01 10 1101 11101	uucu i	within the delimition (	If the Envelope Pavioad (Fi	P\?	j	- Andrew-	
٥.	gases into the	anulua 1e Far	th's atmosphere or s	naterial other than propulsion	on system exh	aust or inert		х
4.	Would launc	h of th	ne candidate spaceci	raft suggest the potential fo				
	ou babile ue	ailli ai	iu saiety?			1	-	x
5.	Would the ca	andida	te spacecraft utilize	a laser system that does no	ot meet the rea	uiremente		
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	operations,	provide	a copy of the hazar	d evaluation and written sa	afety precautior	ns (NPG	***************************************	X
	871530					1		SILVE SI SILVE SI SILVE SI SI SI SI SI SI SI SI SI SI SI SI SI
6.	01 10.01.			1 nothogonia refere		~		- 40
6.	Would the ca	andida	te spacecraft contair	n pathogenic microorganisn uce disease or toxins hazar	ns (including ba	acteria,	Village of the state of the sta	X

NASA Routine Payload Checklist (2 of 2)

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~~~~	lex	Date of Launch: 3/25/03
PROJECT CONTACT:	Mr. Frank Snow	PHONE NUMBER: 301-286-7494 MAILSTOP: 410
PROJECT START DATE:	12/5/97 Phase A	PROJECT LOCATION: GSFC Explorer's Office: Launch Site: KSC
PROJECT DESCRIPTION	Perform 1 <sup>st</sup> ultraviolet all-s Redshift Range 0 <z<2< td=""><td>sky Imaging Survey and Map History of Star Formation over</td></z<2<>	sky Imaging Survey and Map History of Star Formation over

r.	(	OTHER ENVIRONMENTAL ISSUES:	YES	NC
	1.	Would the candidate spacecraft have the potential for substantial effects on the environment outside the United States?		X
2	2.	Would launch and operation of the candidate spacecraft have the potential to create substantial public controversy related to environmental issues?		X
Con	nn	nents:	<u> </u>	<u> </u>

Table 1: Launch Vehicles and Launch Pads

Launch Vehicle	Eastern Range	Western Range
	(CCAFS Launch Complexes)	(VAFB Space Launch Complexes)
Atlas IIA & AS	LC-36	SLC-3
Atlas IIIA & B	LC-36	SLC-3
Atlas V Family	LC-41	SLC-3
Delta II Family	LC-17	SLC-2
Delta III	LC-17	N/A
Delta IV Family	LC-37	SLC-6
Athena I & II	LC-46 or -20	California Spaceport
Taurus	LC-46 0r -20	SLC-576E
Titan II	N/A	SLC-4W
Pegasus XL	CCAFS skidstrip KSC SLF	VAFB airfield

Table 2: Summary of Envelope Spacecraft Subsystems and Envelope Payload Characteristics (EPC)

Structure	Unlimited: aluminum, magnesium, carbon resin composites, and titanium Limited: beryllium [50 kg (110 lb)]
Propulsion	Mono- and bipropellant fuel; 1000 kg (2200 lb) (hydrazine); 1000 kg (2200 lb) (monomethyhydrazine) Bipropellant oxidizer; 1200 kg (2640 lb) (nitrogen tetroxide) Ion-electric fuel; 500 kg (1100 lb) (Xenon) SRM; 600 kg (1320 lb) (AP)-based solid propellant
Communications	Various 10-100 W (RF) transmitters
Power	Solar cells; 150 A-Hr (Ni-H <sub>2</sub> ) battery; 300 A-Hr (LiSOC) battery; 150 A-Hr (NiCd) battery
Science instruments	10 kW radar ANSI safe lasers (Section 4.1.2.1.3)
Other	Class C EEDs for mechanical systems deployment Radioisotopes limited to quantities that are approved for launch by NASA Nuclear Flight Safety Assurance Manager Propulsion system exhaust and inert gas venting

### Summary of GALEX Subsystems

Structural Materials	Aluminum, graphite epoxy composites, invar telescope components, fused silica optics, fiberglass, MLI Blankets, Stainless Steel Hardware
Propulsion	No Propulsion Subsystem
Communications	5 Watt S-band Transmitter, 6 Watt X-band Transmitter
Power	GaAs Solar Arrays 11 CPV Nickel Hydrogen (16 Amp hour Battery)
Science	Near UV and Far UV Detectors,
instruments	Telescope: F 3.0 Meter 0.5 Meter Aperture
Other	Satellite Mass 277.2 Kg
(include dimensions and weight of s/c)	Overall Dimensions:
<i>Q</i> ,	1.97 Meters Tall,
	1.12 Meter Diameter with Solar Arrays Stowed. 2.79 Meters Wide with Solar Arrays Deployed

## **NEPA Environmental Checklist**

1.	Project/Program Galex / GSFC Code 410 Small Explorer's Program Office
2.	
3.	Formulation Process (Phase A/B): Phase A: 12/0519/97; Phase B: 05/04/1998 Implementation Process (Phase C/D): Phase C: 02/23/1999 Launch Date: 03/25/2003 Other Milestone Dates: N/A
4.	Current status  Galex was transported to the KSC Launch Site for final checkouts on 02/02/2003.  The project is on schedule for the 03/25/03 Launch Date.
5.	Project Description  a. Purpose/Need: Galex will perform the first Ultraviolet All-Sky Imaging Survey and will map the history of the universe over the Redshift Range 0 <z<2. a="" and="" answer="" database="" distant="" elements="" formation="" galaxies,="" galaxies.<="" history="" images="" imaging="" metal="" nearby="" of="" originated.="" powerful="" produce="" production="" questions="" resulting="" science="" seen="" spectra="" spectroscopy="" star="" stars="" statistically="" surveys="" td="" the="" today="" uv="" when="" where="" will=""></z<2.>
	b. Spacecraft/Instruments: Orbital Sciences Corporation, Space Sciences Group, is the spacecraft vendor. The three-axis stabilized spacecraft will be in a 28.5 deg inclination/690 km LEO orbit. S/C mass is 277.2kg; S/C power output is 290 watts. Additional S/C characteristics include: Inertial Pointing ACS, 11-16Ahr Common-Pressure-Vessel (CPV) Nickel-Hydrogen batteries, GaAs Fixed Solar Arrays with 3m² total area, 4 reaction wheels, a star tracker, and sun sensors. There is no propulsion, consumables, or cryogenics. All subsystems are made of materials and components commonly used in the space industry. All S/C and instrument materials and quantities are within the Envelope Payload Characteristics.

The Instrument was primarily developed by Cal Tech and JPL. It utilizes a 50cm aperture optical telescope with Al and MgF<sub>2</sub> coatings, MgF<sub>2</sub> and Fused Silica Detector Windows, CsI and CsTe Detector Photo cathodes, an ion-etched, fused silica aspheric astigmatism corrector, a dichroic beam splitter, and a rotating grism wheel. All materials and components are commonly used in the space industry.

c. Launch Vehicle: <u>The Launch Vehicle is a Pegasus XL</u>, designed and <u>manufactured by Orbital Sciences Corporation</u>, <u>Launch Services Group (OSC LSG)</u>.

CCAFS skidstrip

d. Launch Site:

e. Alternatives (to or for the mission): None
f. NASA's Involvement/Responsibility: Galex is a GSFC, Code 410 Small Explorer's Program Office Mission. Galex is a Principle Investigator (PI) mission type. The Explorer's Program Office and the Galex Mission Manager responsible for the management of Galex project development, launch, and on-orbit checkout.
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g. Participants/Locations: California Institute of Technology (Cal Tech) is the Galex PI, with overall responsibility for delivering the Galex S/C and associated mission science to GSFC. Cal Tech developed the instrument science, provided test support, and will conduct the mission operations, including science operations, data analysis, and science analysis. JPL, under contract to Cal Tech, provides overall project management and implementation. They also developed the Galex instrument. The S/C vendor, also under contract to Cal Tech, is Orbital Sciences Corporation Space Science Group (OSC SSG), in Dulles, Virginia. They developed the S/C bus, performed integration & testing, developed L/V interfaces, and are responsible for the Ground Data Systems and conducting Mission Operations. The Mission Operations Center is also located at the OSC facility in Dulles, Va.

h. Mission Life: Baselined Mission Life is 28 Months.

i. End of mission, Re-entry: 4 year useful mission life. The Galex Orbital Debris Analysis indicates both the L/V and the S/C exceed the <25-year maximum on lifetime orbit guidelines. The S/C will re-enter in 56 years. The Pegasus third stage will re-enter in 54 years. The Galex project has investigated both lower and higher orbits to reduce either the orbit lifetime or park the satellite in a long duration orbit. Neither option is feasible. Lower orbits would result in curtailing the science mission. 690km was selected as the lowest orbit consistent with the nominal science mission. Orbits higher than 800km also resulted in reduced science. The Pegasus XL capability is also limited to about 800 km for the Galex Satellite Mass. This non-conformance is the only non-conformance in the Galex

Orbital Debris Analysis. Code Q recommended acceptance of the Galex Orbital Debris Analysis, with the one identified non-conformance.

6.	Is there anything controversial about the mission?
7.	Is there anything unique, unusual, exotic about the mission, spacecraft, and instruments? $\frac{No}{-\infty}$
8.	Is there any environmental documentation for spacecraft, launch vehicle (NEPA or EO12114)?  Spacecraft - No
	Launch Vehicle - Yes, the Pegasus XL has a previously documented and approved Environmental Impact Assessment.
9.	Has an orbital debris analysis been completed? (If yes, what is returning to earth?) Yes; see 5i above. The analysis concludes the Pegasus third stage fails to burn up completely and leaves debris over a 1.56m² area. The S/C also fails to burn up completely and leaves debris over a 5.637m² area. Both projections meet the less than 8m² requirement.
10.	Has an Air Force Form 813 been completed? (Please attach copy)
11.	Does the mission include or involve:  Check all that apply. If uncertain indicate with a "?"  For all that apply provide an explanation. Use the additional space below if needed.
	L/V a. Fuels Solid Propellants Only b. Radioactive Material
	L/V c. Explosives Destruct Ordnance
	d. Chemicals
	L/V e. Hazardous Materials/Substances Destruct Ordnance
	S/C f. Lasers (Class, Earth Pointing) Contained, Non-Farth Pointing
	g. Disease Producing Pathogenic Microorganisms
	n. Construction/Modification of a Facility
	L/V i. Discharges of any substances into air, water, or soil Rocket Exhaust
	L/V j. Generation/Use/Storage/Disposal of Toxic or Hazardous Substances  Destruct Ordnance

	L/V; S/C k. Generation of Hazardous Wastes Normal Launch Site Processing Waste
	Products (alcohol wipes, etc); KSC is responsible for proper hazardous material management and disposal.
	L/V l. Generation of High Noise Levels _L-1011 Take-off on KSC Skid Strip
,	m. Sample Return to Earth
	L/V; S/C n. Generation of Ionizing or Nonionizing Radiation Both produce non-ionizing radiation. Sources are transmitters and lasers.
	o. Impact on Local Social or Economic Conditions
	p. Removal of Vegetation or Destruction of Habitat
•	q. Impact/Affect on Minority or Low Income Populations
•	r. Affect Any Threatened or Endangered Species
-	s. Affect Areas of Historical or Cultural Significance
~	t. New or Foreign Launch Vehicle
_	t. New or Foreign Launch Vehicle
-	u. Other Issues of Potential Environmental Impact
Project	ant Snow  Manager, Code  3/20/03  Date
Explana L/V – all limitation	ations  L/V affirmative responses are enveloped within material, quantity, and power as already documented and approved in previous Pegasus XL NEPA documents.
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Laser rec	alex lasers are contained and non-earth pointing (not applicable to class III or IV quirements and restrictions). All non-ionizing exposure levels are below MPE's d in the Routine Payload EA guidelines (enveloped within NASA and ANSI ents).